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25X1A S E C R E T 241825Z DEC 69 CITE 25X1A 25×1A IDEAL IST 25X1A REFS: A. (NOTAL) 25X1常 В. (IN 37012X (IN 38595) 25X1A SUBJECT: FLIGHT TEST PLAN 1. RE REF A SCHEDULE: APPROVED TO FLY MAGIC PAINT TEST FLIGHT ON 7 JAN 1970 USING ART 383 AS TEST AIRCRAFT, ART 055 AS HIGH CHASE AND T-33 AS LOW CHASE. BACKUP DATE FOR TEST IS 8 JAN 1978. 2. APPLICABLE PORTIONS OF HGS GUIDANCE IN PARA 2 REF B APPLY FOR THIS TEST. 3. ART 383 PREPARATION AS FOLLOWS: 25X1 A. PAINTING WILL BE ACCOMPLISHED AT 25X1A BY PAINTERS ON 29 - 31 DEC. OLD TEST PAINT SURFACES WILL BE STRIPPED OFF PRIOR TO APPLYING NEW PRIMERS AND MAGIC PAINTS. CARE MUST BE TAKEN TO PROTECT NEW SURFACES PRIOR TO FLIGHT. B. FOLG COMBINATIONS OF HCP FORMULATIONS AND PRIMERS TO BE TESTED:

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- (1) NEW HIGH TEMP (PLUS 14 DEGREES F) OVER WHITE UNDER COAT.
- (2) NEW HIGH TEMP (ALUS 14 DEGREES F) OVER YELLOW.
- (3) NEW HIGH TEMP (PLUS 14 DEGREES F) OVER GREEN.
- (4) LOW TEMP (MINUS 8 DEGREES F) OVER WHITE.
- (5) LOW TEMP (MINUS & DEGREES F) OVER YELLOW.
- (6) LOW TEMP (MINUS & DEGREES F) OVER GREEN.
- 25X1A c. PAINT SCHEME:
 - (1) TEST SURFACES WILL BE AS AGREED TO IN TELECON

23 DEC 1969, I.E., THE AREA AFT OF 48 PERCENT BEAM

- OUTBOARD OF WING STATION 300 TO WING TIP, TOP AND
- _ BOTTOM, AND NOT INCLUDING THE SKID AREA. ADDITIONALLY,
 - THE TOP AND BOTTOM OF THE WING FLAP SURFACES
- INBOARD FROM THE EXISTING REFERENCE COLOR SPECTRA
- WILL BE TREATED. THE SIZE OF THE FLAP TEST AREA WILL BE
 - DETERMINED BY THE REMAINING AVAILABLE MAGIC PAINT
- SUPPLY. PREVIOUSLY TREATED AREA ON THE VERTICAL
- STAB WILL NOT BE REPAINTED AND WILL BE USED FOR AGING
 - COMPARISON TO PREVIOUS TESTS.
 - (2) EACH WING WILL HAVE THREE TEST AREAS (TOP AND

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BOTTOM); TWO PANELS OF APPROX EQUAL APEA WILL BE OUTBOARD OF EXISTING COLOR SPECTRA AND SEPARATED BY A 6 INCH BAND OF VELVET BLUE; THE OTHER AREA WILL BE INBOARD OF THE COLOR SPECTRA ON THE FLAP SURFACE. THE NON-TREATED WING SURFACES FORWARD OF 48 PER CENT

- BEAM OUTBOARD OF WING STATION 320 INCLUDING THE SKITS, WILL BE PAINTED VELVET BLUE FOR CONTRAST TO THE TEST PANELS.
 - (3) THE OUTBOARD PANEL ON EACH WING WILL USE A YELLOW UNDERCOAT.
 - (4) THE CENTER PANEL ON EACH WING WILL USE A GREEN UNDERCOAT.
 - (5) THE INBOARD (FLAP) PANEL WILL USE A WHITE UNDERCOAT.
 - (6) THE NEW HIGH TEMP FORMULATIONS (PARA 38) WILL BE APPLIED TO THE LEFT WING.
 - (7) THE LOW TEMP FORMULATIONS (PARA 3B) WILL BE APPLIED TO THE RIGHT WING.
- D. THE REFERENCE COLOR SPECTRA WILL BE USED AS FOR PREVIOUS FLIGHT.
 - E. SMALL DETACHABLE PANELS TO BE TREATED WITH

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| | EACH OF THE TEST FORMULATIONS WILL BE ATTACHED SO THE PILOT | r. |
| | MAY OBSERVE THEM IN FLIGHT. | |
| ~ | F. INSTRUMENTATION- REQUEST THERMOCOPULE | |
| | INSTALLATION MODIFIED TO MEASURE TEMPERATURE ON EACH OF THE | |
| | FOUR OUTBOARD TEST SURFACES TOP AND BOTTOM. INSTALLATION WILL | |
| • | BE CALIBRATED AND TESTED PRIOR TO FLIGHT. | |
| . . | 4. HIGH PORTION TEST PROCEDURES - ART 383 AND 855 WILL TAKE | |
| | OFF AND CLIMB IN LOOSE VISUAL FORMATION TO OPERATIONAL ALTITUDE | |
| | (LEVEL OFF DESTRED AT 70M, IF FEASIBLE). DURING CLIMB PILOTS WILL | |
| | RECORD ANY UNUSUAL COLOR PHENOMENA OBSERVED. AT LEVEL OFF THE | |
| | HIGH CHASE (855) WILL ACCOMPLISH VISUAL CHECKS, PHOTOGRAPHY | |
| - | AND DATA RECORDING ON THE SHIPS RECORDER SYSTEM. DESIRE | |
| | ADDITIONAL VISUAL CHECKS FROM APPROX 1 NM ASTERN ABOVE AND BELOW | |
| | TO RECORD ANY DIFFERENCES NOTED FROM CLOSE UP OBSERVATION. | |
| • | WHEN HIGH PORTION OF TEST COMPLETE, AIRCRAFT WILL BREAK OFF | |
| | FORMATION AND CONTINUE ON BRIEFED PHOTO MISSION. FORMATION | |
| | PROCEDURES AND SAFETY CRITERIA WILL BE AT COMDR | 25X1A |
| • | DISCRETION. FLYING SAFETY IS PARAMOUNT AND WILL NOT BE COMPROMISED | • |
| ; , | TO ACCOMPLISH TEST OBJECTIVES. | |
| ' | 5. LOW PORTION TEST PROCEDURES - DURING TERMINAL DESCENT, | |

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| | THE TEST AIRCRAFT WILL RENDEZVOUS WITH T-33 AT FL 388 - 350 WITH THE | , |
| | EXACT ALTITUDE DETERMINED BY THE COLDEST FORECAST TEMP AND | e st . |
| | ALT COMBINATION WITHIN CPABILITY OF BOTH AIRCRAFT. DATA | : |
| _ | COLLECTION, RENDEZVOUS, JOINUP, AND FORMATION SAFETY PROCEDURES | ÷ |
| | FOR THE PORTION OF THE TEST WILL BE AS PREVIOUSLY ESTABLISHED. | 1 |
| • | 6. VISUAL OBSERVATIONS - IN THE INTEREST OF SEEKING UNIFORM | |
| 25X1 <u>A</u> | TEST RESULTS, REQUEST SAME PILOTS PARTICIPATE IN THIS | *9 |
| | TEST AS FOR PREVIOUS TESTS. | |
| • | 7. PHOTOGRAPHY - SAME REQUIREMENTS AS FOR PREVIOUS TESTS. | |
| | PARA 7. REF C APPLIES FOR THIS TEST. USE OF TELEPHOTO LENSE AND | |
| | DOUBLE FILTER ARRANGEMENT COMPRISED OF A 90 - 750 MICRON DICHROIC | |
| • | WITH A HARRISON HAVING A DENSITY ABOUT HALF THAT OF WRATTEN 85, | |
| • | IS DESIRED. | 25X1 <i>A</i> |
| 25X1A | 8. TECHNICAL COORDINATOR MOT PLANNING TO OBSER | |
| • | THIS TEST. TELECON COORDINATION IF PROBLEMS ARISE, IS ENCOURAGED. | |
| | 9. BARRING UNFORESEEN PROBLEMS, ANTICIPATE THAT THIS WILL | 25X1/ |
| _ | BE LAST PHASE I TEST. REQUEST PROVIDE | 25X1A |
| • | WITH FINAL EVALUATION SUMMARY OF PHASE I TEST PROGRAM TO INCLUDE | |
| - | RECOMMENDATIONS FOR FURTHER DEVELOPMENT AND OPERATIONAL | i |
| - | APPLICATION TO 11-08 | |

END OF MSG